

2018 International Conference on Thermoelectrics

RTGS: THE ENDURING AND THE FUTURE

David Woerner



Large, strategic NASA missions

A recent report, by the US National Academy of Sciences, Engineering, and Medicine, entitled, *Powering Science: NASA's Large Strategic Science Missions* (2017), asserted,

"Large, strategic missions "produce tremendous science returns and are a foundation of the global reputation of NASA and the U.S. space program."



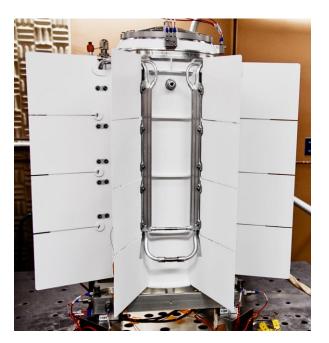
- Used by NASA missions of various types for over 50 years
 - Pioneer
 - Viking
 - Voyager
 - Galileo
 - Ulysses
 - Cassini
 - Pluto New Horizons
 - Curiosity



Multi-Hundred Watt – Radioisotope Thermoelectric Generator (MHW–RTG) (Voyager)



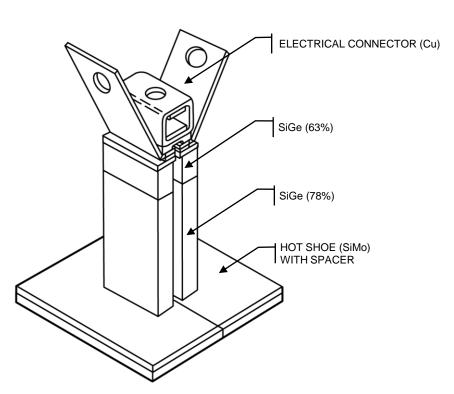
General Purpose Heat Source – Radioisotope Thermoelectric Generator (GPHS–RTG) (Galileo, Ulysses, Cassini, Pluto New Horizons)



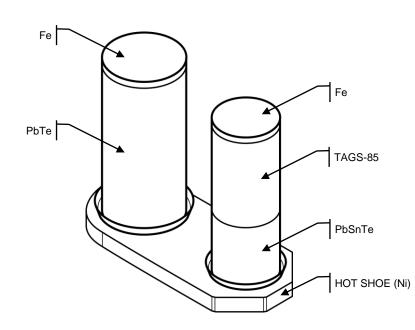
Multi-Mission Radioisotope
Thermoelectric Generator (MMRTG)
(Curiosity)

- NASA was flying 11 RTGs if you count the recently ended Cassini mission
 - Voyager, 3 MHW RTGs/spacecraft
 - Cassini, 3 GPHS-RTGs
 - PNH, 1 GPHS-RTG
 - Curiosity, 1 MMRTG

Multi-Hundred Watt – Radioisotope Thermoelectric Generator (MHW–RTG) (Voyager) General Purpose Heat Source – Radioisotope Thermoelectric Generator (GPHS–RTG) (Galileo, Ulysses, Cassini, Pluto New Horizons) Multi-Mission Radioisotope
Thermoelectric Generator (MMRTG)
(Curiosity)



Silicon-Germanium Unicouple

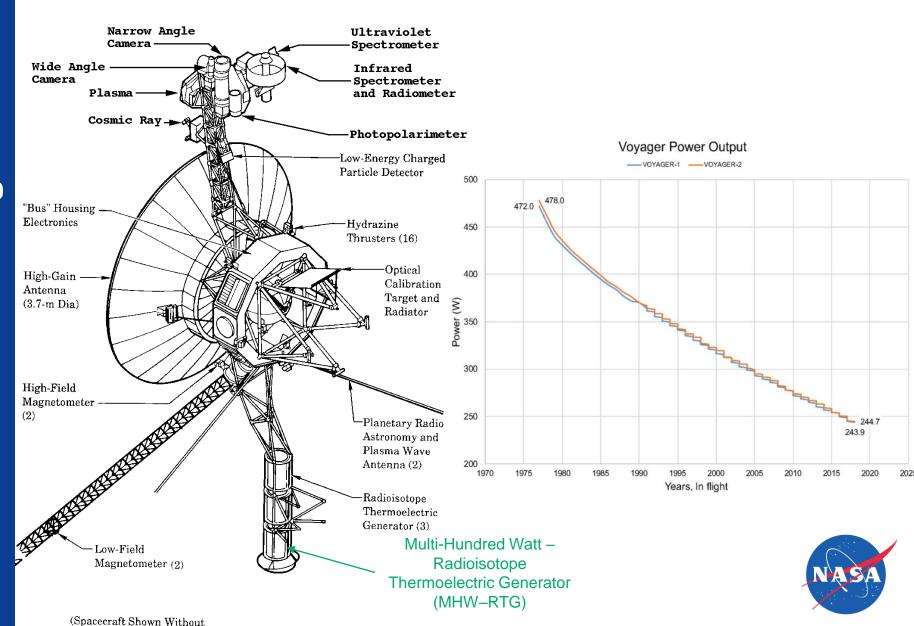


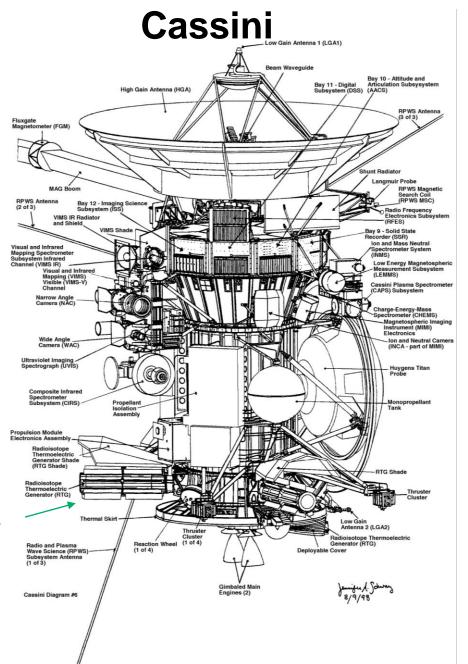
Lead-Telluride-TAGS Unicouple



Thermal Blankets for Clarity)

Voyager 1 and 2

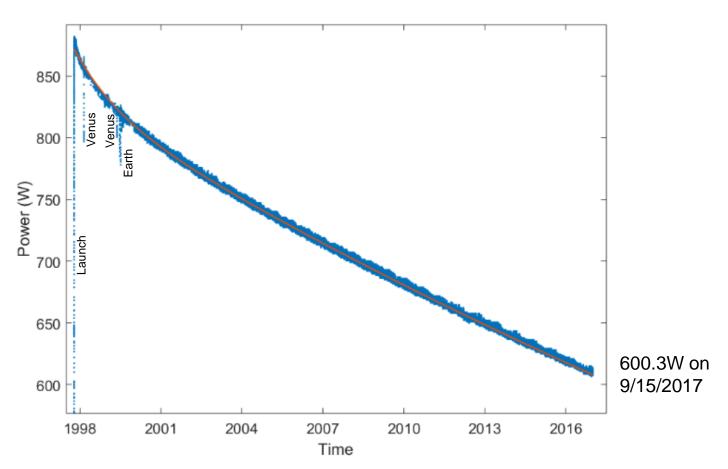




General Purpose Heat Source – Radioisotope Thermoelectric Generator (GPHS–RTG)



Cassini



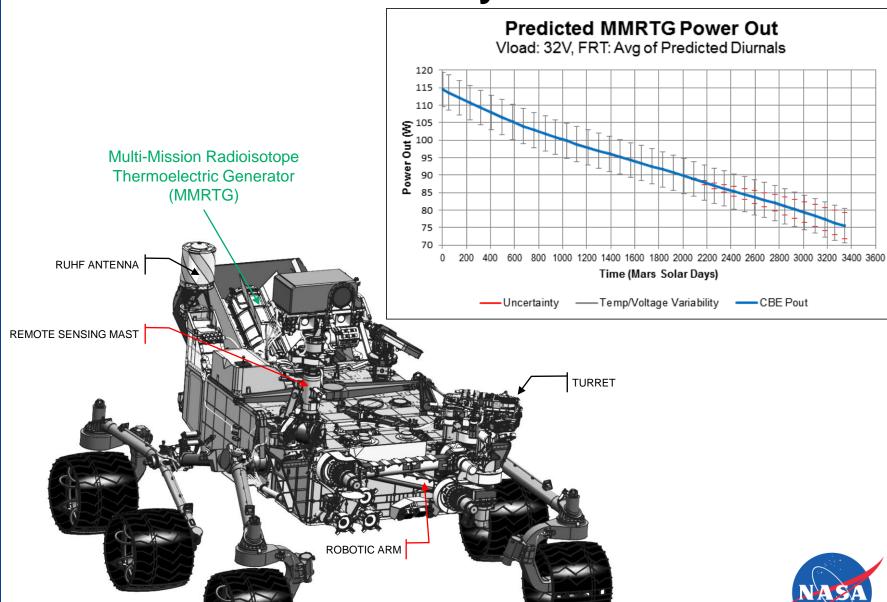
Complete Cassini telemetry data between launch and 12/31/2016



Pluto New Horizons Forward low-gain antenna High-gain antenna Medium-gain antenna REX **PEPSSI** Radioisotope General Purpose Heat Source thermoelectric - Radioisotope Thermoelectric generator (RTG) SWAP Generator (GPHS-RTG) **Thrusters** RTG OUTPUT POWER PREDICTION →NH RTG Power ■■ NH RTG Last Year Predictions Expon. (NH RTG Last Year) 245 Ralph Star Trackers 240 **Thrusters** 235 230 225 220 215 210 Pluto POWER (WATTS) 205 200 **KBO** 195 190 185 180 175 170 165 ~82AU 160 155 150 10 12 14 16 20 22 24 26

Pre-decision TIME (YEARS FROM LAUNCH)

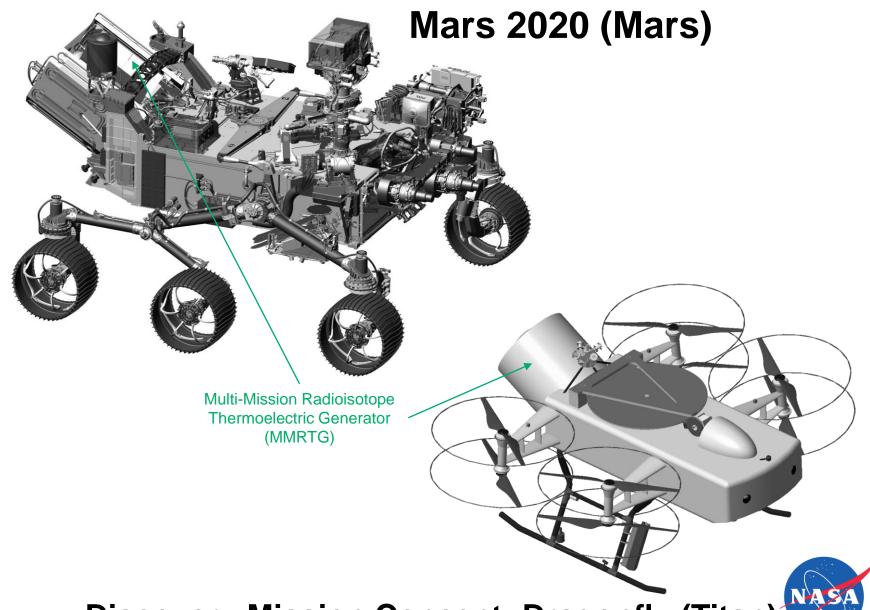
Curiosity





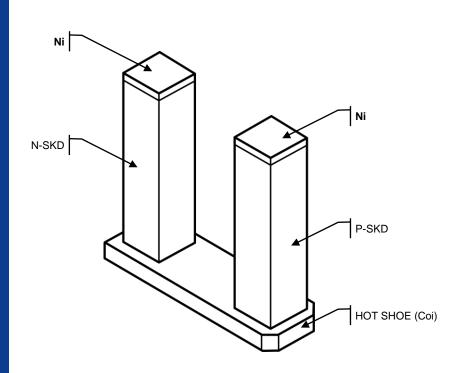
The Future

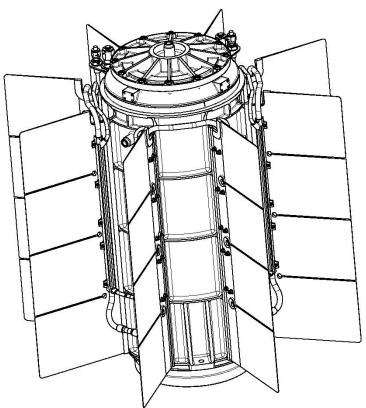




Discovery Mission Concept: Dragonfly (Titan)

The potential enhanced MMRTG

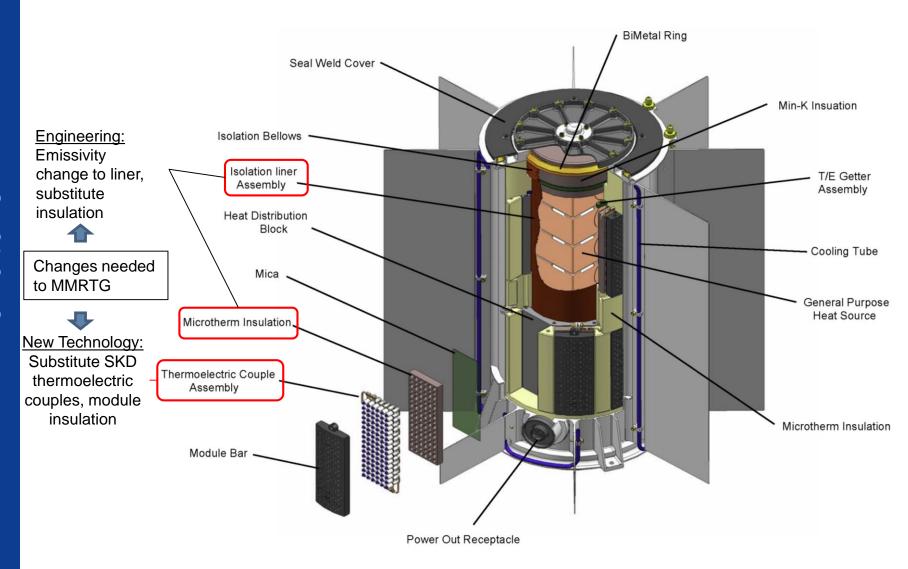




Enhanced Multi-Mission Radioisotope Thermoelectric Generator (eMMRTG)



The potential enhanced MMRTG





The Next-Generation RTG Study

Study Lead: David Woerner

D.:-	Delasta	IDI
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Pradeep	Bhandari	JPL
Chester	Borden	JPL
Sevan	Chanakain	JPL
Alan	Didion	JPL
Fivos	Drymiotis	JPL
John	Elliot	JPL
Jean-Pierre	Fleurial	JPL
Terry	Hendricks	JPL
Insoo	Jun	JPL
Reh	Kim	JPL
Damon	Landau	JPL
Young	Lee	JPL
Hari	Nayar	JPL
David	Neff	JPL
Arora	Nitin	JPL
Knut	Oxnevad	JPL
Martin	Ratliff	JPL
Timothy	Shirey	JPL
David	Woerner	JPL
Kevin	Yu	JPL
Chadwick	Barklay	UDRI
Dirk	Cairns-Gallimore	DOE
Steve	Johnson	DOE INL
Tom	Spilker	Independent

Mike	Amato	GSFC
David	Batchelor	GSFC
Bob	Beaman	GSFC
Porfy	Beltran	GSFC
Kim	Brown	GSFC
Jacob	Burke	GSFC
Jacob	Englander	GSFC
Matthew	Garrison	GSFC
John	Godfrey	GSFC
Kyle	Hughes	GSFC
Frank	Kirchman	GSFC
Jeremy	Knittel	GSFC
Blake	Lorenz	GSFC
Paul	Mason	GSFC
Anthony	Nicoletti	GSFC
Anthony	Nicoletti	GSFC
Dave	Palace	GSFC
Daniel	Ramspacher	GSFC
David	Robinson	GSFC
Terry	Smith	GSFC
James	Sturm	GSFC

Amee	Bogner	GRC
Elizabeth	Turnbull	GRC
Evan	Roelke	GRC
James	Fittje	GRC
Jeff	Woytach	GRC
John	Gyekenyesi	GRC
June	Zakrajsek	GRC
Justin	Walsh	GRC
Mike	Martini	GRC
Paul	Schmitz	GRC
Robert	Jones	GRC
Steve	McCarty	GRC
Steve	Oleson	GRC
Tom	Packard	GRC
Tom	Parkey	GRC
Tony	Colozza	GRC
Reed	Cheryl	APL
Ken	Hibbard	APL
Ralph	Lorenz	APL
Procktor	Louise	APL
Paul	Michael	APL
Paul	Ostdiek	APL
Ostdiek	Paul	APL
Dennis	Woodfork	APL

APL – Applied Physics Lab DOE – Department of Energy GRC – Genn Research Center GSFC – Goddard Spaceflight Center INL – Idaho national Lab



Next-Gen RTG Study completed in 2017

Was motivated by the need for larger RTGs than presently available or near-term improvements

- Serve NASA for 2-3 decades to come
- To address the needs of future Decadal Survey missions
 - ✓ An RTG that would be useful across the Solar System
 - ✓ An RTG that maximizes the types of missions: flyby, orbit, land, rove, boats, submersibles, balloons
 - ✓ An RTG that has reasonable development risks and timeline



"Flown"

Approach For The Study Final report **Destinations (63)** MMRTG/ **GPHS-RTG** Visit rps.nasa.gov > (Visited or suggested in Decadal Surveys) **eMMRTG** Req. Resources > Reports Req. Neptune Europa Venus "Gas" "Ocean" "Ice" Requirements **RTG Concepts** Performance - General Purpose for JPL max. fit **Physical** - Specialized RTGs **TE Technologies** Structural for significant UDRI, JPL niches Environmental - Timeline - Literature search - Lab data - Screen materials - Model couples Launch Vehicles (4) Spacecraft/Missions (99) / Mission Types (Flown and Studied) SLS (1 A and Delta IV Atlas V (541) Venus Rover Cassini B) Titan Submarine Heavy Launched: MSL (Surface) Titan IV B (Orbiter) Potential (Subsurface) Potential "Suggested"

Launcher

Launcher

Launched:

Cassini

"Suggested"

Thermoelectric Technology

Configuration	# Segments	~Couple Efficiency at Tcj 450K	Mate	RL erials p	TRL of Configuration	~ Generator Efficiency (16 GPHSs)
1	3 Element	17	9/2/2	9/2.5/3.5	1	14.8
2	3 Element	15	9/ <mark>2</mark> /3.5	9/2.5/3.5	1	13.6
3*	3 Element	16	9/4/ <mark>2</mark>	9/4/3.5	2	13.9
4*	3 Element	14	9/4/3.5	9/4/3.5	2.5	12.7
10	1 Element	14	2	3.5	2	12.1
11	1 Element	11	3.5	3.5	3.5	10
14	2 Element	14	9/2	9/3.5	2.5	12.6
21	2 Element	12	9/3.5	9/3.5	2.5	10.6

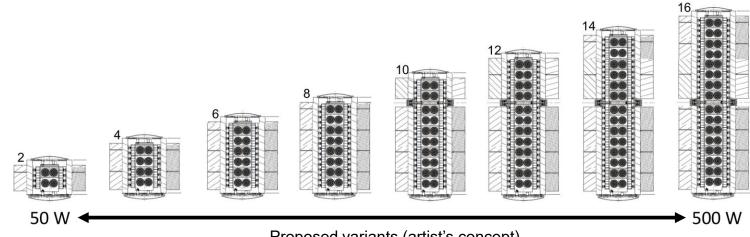
^{*} Contains SKD

Choose ~4 (1, 3, 4, 14)
Prefer 1- and 2-segment couple configurations



Concepts

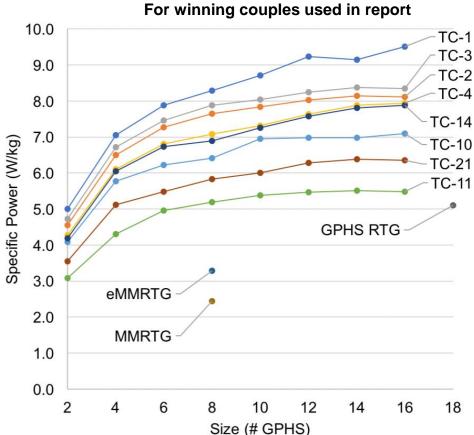
- 3 surviving Types of Next-Gen RTG Concepts:
 - Vacuum Only
 - Single-point design using single couple technology
 - Modular design using multicouple technology
 - Vacuum and Atmosphere
 - > Hybrid
- Variants: 2, 4, 6, 8, 10, 12, 14, and 16 GPHS variants



Specific Power

- Figure displays the BOL specific power estimates for the modular concepts, in W/kg, for the various size options of each TEC selection.
- Nearly all options exceed the specific power of both the MMRTG and eMMRTG
- The greater power output of the larger configurations may eliminate the need for multiple RTGs on certain missions, saving significant mass and fuel resources.
- The specific power of a generator increases with the number of GPHS used, demonstrating that a single larger NG-RTG is more mass efficient than multiple smaller NG-RTG concepts.

BOL Specific Power





Summary Recommendation

- Next-Generation RTG baseline design decisions:
 - Vacuum-only
 - Modular
 - 16 GPHS (largest RTG variant)
 - P_{BOM} = 400-500 W_e (largest RTG variant)
 - Mass goal of < 60 kg (largest RTG variant)
 - Degradation rate < 1.9 % per yr on average and including fuel degradation
 - System designed to be upgraded with new TC's as technology matures



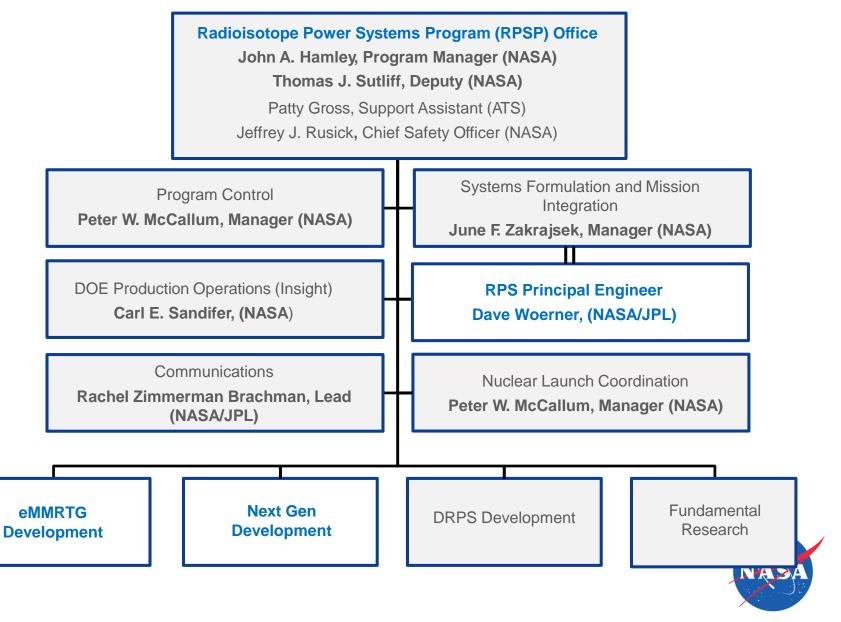
A Sea-change

/ˈsē ˌCHānj/ noun

1. a profound or notable transformation.



RPS Program New Organization



Development Framework

The future

	RTG	Converter Couple Technology	System Integrator	JPL role	Where is the Tech now?	
D	<i>MHW-RTG</i> OE Funded	Si-Ge unicouple from RCA	GE Astrospace (now LMA)	Converter Tech SE, V&V, LPP	Not available	
D	GPHS-RTG OE Funded	RCA Si-Ge tech transferred to GE Astrospace	GE Astrospace (now LMA)	Converter Tech SE, V&V, LPP	Not available	
\ \ \	MMRTG IASA Funded	PbTe/TAGS couples at TESI, based on previous SNAP RTG Tech by TESI	Aerojet Rocketdyne (AR) assisted by TESI	Converter Tech SE, V&V, LPP	Current only "off-the-shelf" Tech	Past
N	eMMRTG ASA Funded	SKD couple tech at JPL transferred to TESI	Aerojet Rocketdyne (AR) assisted by TESI	Tech transfer & maturation, Converter Tech SE, V&V, LPP	In Tech Maturation phase, led by NASA/JPL (TTDP); Potential transition to DOE-led flight system development phase in FY19	Future
N	Next-Gen RTG ASA Funded	Segmented TE tech at JPL to be transferred to industry (competed/directed)	Unknown	Tech transfer & maturation, Converter Tech SE, V&V, LPP	In Tech Advancement phase, led by NASA/JPL (TTDP); Potential transition to Tech Maturation phase in FY20, DOE- led flight system development phase in FY23	

Past implementation: technology was located at Industry New implementation: technology located at NASA/JPL

Summary of Current Schedules and Status

eMMRTG

- Almost 6 years of technology maturation complete
- Project instantiated, January 2018
- 7 months to next technology decision gate review
- Complete technology transfer to industry/begin development of qualification generator: 2019
- Deliver qualification unit, unfueled: 2024

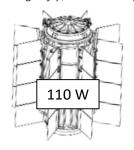
Next-Generation RTG

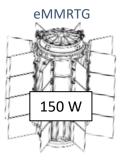
- Project instantiated, March 2018
- Initiated TEC maturation for NG-RTG
- Begin development of engineering unit generator: 2022
- Complete technology transfer to industry: 2023
- Begin development of qualification generator: 2024
- Deliver qualification unit, unfueled: 2028



RTG Comparison

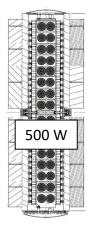
MMRTG (Curiosity, Dragonfly, Mars 2020)







Next-Generation RTG Concept, 16-GPHS version



Power, launch, W	110	150	290 (880)	500
Power, end of life, W	55	91	213 (640)	362
Degradation rate, average	4.8%	2.5%	1.9%	1.9%
Design Life, years	17	17	18	17
# GPHSs	8	8	18	16
Length, m	0.69	0.69	1.14	1.04
Mass, kg	45	44	57	62

Largest variant would be expected to have higher maximum power output, more efficient use of fuel, and a low degradation rate compared with previous RTGs



New thermoelectric materials and technologies and RTG designs are being engineered for potential use on deep space missions for the first time in ~50 years.

It is a spectacular time to be working on thermoelectrics and RTGs

Thank you.

The Final Report of the Next-Generation RTG Study is here: https://rps.nasa.gov/resources/73/next-generation-rtg-study-final-report/?category=reports





jpl.nasa.gov

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